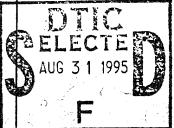
26 MAY 1950

# NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS



REPORT 931

CORRELATION OF CYLINDER-HEAD TEMPERATURES
AND COOLANT HEAT REJECTIONS OF A
MULTICYLINDER, LIQUID-COOLED ENGINE
OF 1710-CUBIC-INCH DISPLACEMENT

By BRUCE T. LUNDIN, JOHN H. POVOLNY, and LOUIS J. CHELKO



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1949

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# AERONAUTIC SYMBOLS

# 1. FUNDAMENTAL AND DERIVED UNITS

-	-	Metric		English					
	Symbol	Unit	Abbrevia- tion	Unit	Abbrevia- tion				
Length Time Force	l t F	metersecondweight of 1 kilogram	m s kg	foot (or mile) second (or hour) weight of 1 pound	ft (or mi) sec (or hr) lb				
Power	<b>P</b> V	horsepower (metric) kilometers per hour meters per second	kph mps	horsepower miles per hour feet per second	hp mph fps				

# 2. GENERAL SYMBOLS

W	Weight=mg Standard acceleration of gravity=9.80665 m/s or 32.1740 ft/sec <sup>2</sup>	Sta	Kinematic viscosity Density (mass per unit volume) Indard density of dry air, 0.12497 kg-m <sup>-4</sup> -s <sup>2</sup> at 15° ( and 760 mm; or 0.002378 lb-ft <sup>-4</sup> sec <sup>2</sup>
m	$Mass = \frac{\eta}{q}$	Spe	ecific weight of "standard" air, 1.2255 kg/m <sup>3</sup> o
I	Moment of inertia= $mk^2$ . (Indicate axis of	of C	0.07651 lb/cu ft
μ	radius of gyration k by proper subscript.) Coefficient of viscosity		
	3. AERODYN	VAMIC	SYMBOLS
8	Area	$i_{w}$	Angle of setting of wings (relative to thrust line
S	Area of wing	- i.	Angle of stabilizer setting (relative to thrus
$\boldsymbol{G}$	Gap	2	line)
b	Span	Q	Resultant moment Resultant angular velocity
<b></b>	Chord	/ :	
$\boldsymbol{A}$	Aspect ratio, $\frac{b^2}{S}$	R	Reynolds number, $\rho \frac{Vl}{\mu}$ where l is a linear dimen
v	True air speed		sion (e.g., for an airfoil of 1.0 ft chord, 100 mph
_	Dynamic pressure, $\frac{1}{2}\rho V^2$		standard pressure at 15° C, the corresponding
q	Dynamic pressure, $\frac{7}{2}p^{\nu}$		Reynolds number is 935,400; or for an airfoi
$\boldsymbol{L}$	Lift, absolute coefficient $C_L = \frac{1}{qS}$	-	of 1.0 m chord, 100 mps, the corresponding
	and the control of th		Reynolds number is 6,865,000) Angle of attack
$\boldsymbol{D}$	Drag, absolute coefficient $C_{\mathcal{Q}} = \frac{D}{qS}$	α	Angle of downwash
		α <sub>a</sub>	Angle of downwash  Angle of attack, infinite aspect ratio
$D_{o}$	Profile drag, absolute coefficient $C_{D_0} = \frac{D_0}{qS}$	$\alpha_i$	Angle of attack, induced
$D_i$	Induced drag, absolute coefficient $C_{D_t} = \frac{D_t}{qS}$	$\alpha_a$	Angle of attack, absolute (measured from zero lift position)
$D_{\mathfrak{p}}$	Parasite drag, absolute coefficient $C_{Dp} = \frac{D_p}{qS}$	γ	Flight-path angle

Cross-wind force, absolute coefficient  $C_c$ =

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# SUMMARY

Data obtained from an extensive investigation of the cooling characteristics of four multicylinder, liquid-cooled engines have been analyzed and a correlation of both the cylinder-head temperatures and the coolant heat rejections with the primary engine and coolant variables was obtained. The method of correlation was previously developed by the NACA from an analysis of the cooling processes involved in a liquid-cooledengine cylinder and is based on the theory of nonboiling, forced-convection heat transfer. The data correlated included engine power outputs from 275 to 1860 brake horsepower; coolant flows from 50 to 320 gallons per minute; coolants varying in composition from 100 percent water to 97 percent ethylene glycol and 3 percent water; and ranges of engine speed, manifold pressure, carburetor-air temperature, fuel-air ratio, exhaust-gas pressure, ignition timing, and coolant temperature. The effect on engine cooling of scale formation on the coolant passages of the engine and of boiling of the coolant under various operating conditions is also discussed.

The results of this analysis indicated that the correlation method is applicable to multicylinder, liquid-cooled engines of the type investigated and permits the prediction of the cylinderhead temperature between the exhaust valves within approximately  $\pm 12^\circ$  F and of the coolant heat rejection with an accuracy of  $\pm 5$  percent for any operating condition within the range of the investigation.

# INTRODUCTION

An investigation of the cooling characteristics of reciprocating aircraft engines is of importance in order to insure satisfactory engine performance at extreme conditions of operation. A considerable amount of data on the cooling characteristics of various air-cooled engines has been published by various investigators but little data have been published on the cooling characteristics of liquid-cooled engines.

An extensive research program to determine the cooling characteristics of liquid-cooled engines was therefore instituted at the NACA Cleveland laboratory in 1943. The initial phase of this program consisted of an investigation conducted on a single-cylinder engine to provide data for a fundamental study of the heat-transfer processes involved. The final results of this investigation are reported in reference 1 in which an analysis, based on the theory of nonboiling forced-convection heat transfer, was made of the cooling

processes in a liquid-cooled engine. This analysis result in a semiempirical method, similar to that presented reference 2 for air-cooled engines, of correlating the cylind head temperatures with the primary engine and coolevariables; and this method was successfully applied to that.

Following the investigation on the single-cylinder eng (reference 1), a comprehensive investigation of the cooli characteristics of a multicylinder engine of 1710-cubic-indisplacement was conducted. The primary data obtain in this investigation are presented in reference 3 in the fo of plots of the cylinder temperatures and the coolant he rejections against the basic engine and coolant variable In order to determine the applicability of the correlati method of reference 1 to a multicylinder engine and obtain in most conveniently applied form a complete formu tion of the principal cooling characteristics of liquid-cool engines, this semiempirical method was employed in slight modified form to correlate both the cylinder-head-tempature data and the coolant-heat-rejection data of reference with the primary engine and coolant variables. The resu of both of these correlations as well as examples of th application to a typical problem are presented herein.

The data used in the correlations presented in this represented wide ranges of engine and coolant conditions including engine power outputs from 275 to 1860 brake horsepower coolant flows from 50 to 320 gallons per minute, and coolar composed of ethylene glycol—water mixtures ranging composition from 100 percent water to 97 percent ethyle glycol and 3 percent water.

# APPARATUS AND PROCEDURE

The data used in this analysis were obtained from V-17 engines set up on a dynamometer stand and are present in curve form in reference 3. The data used in the corelation of cylinder-head temperatures were obtained from the engines that are designated engines A, B, C, and D reference 3 and herein. Data for the correlation of coolar heat rejections were obtained from engine D only; data from the engines A, B, and C are not included for this correlation because the experimental technique used for these engines was not sufficiently refined to provide heat-rejection data the accuracy required for this analysis. The engine mode used are 12-cylinder, liquid-cooled, V-type engines with displacement of 1710 cubic inches, a 5.5-inch bore, and

6.0-inch stroke. The compression ratio is 6.65 and the engines are fitted with single-stage gear-driven superchargers having a gear ratio of 9.6:1 and an impeller diameter of 9.5 inches. The standard ignition system is timed to fire the intake spark plugs 28° B.T.C. and the exhaust spark plugs 34° B.T.C. The valve overlap extends over a period of time equivalent to 74° rotation of the crankshaft.

The cylinder-head temperatures were measured by ironconstantan thermocouples installed in each cylinder between the exhaust valves, between the intake valves, and in the exhaust spark-plug boss at the locations shown in figure 1 and according to the method described in reference 3.

A schematic diagram of the cooling system is shown in figure 2. Copper-constantan thermocouples, differentially connected to a portable precision-type potentiometer, were installed at the locations shown in figure 2 for the purpose of measuring the coolant temperature rise across the engine and the coolant cooling-water temperature rise across the coolers. The coolant flow was measured by means of a venturi installed in the main coolant line and the coolant coolingwater flow was measured with a calibrated rotameter installed in the cooling-water line. Further details of the instrumentation and a description of the general setup and auxiliary equipment are given in reference 3.

A summary of the engine and coolant conditions covered by the data used for both the cylinder-head-temperature and heat-rejection analyses is given in table I. In order to isolate the effect of the engine and coolant variables on both the cylinder-head temperatures and the coolant heat rejections, one of the conditions was varied in each of the tests, while, in general, all the others were held constant. The tests on engine A covered typical engine operating conditions varying from cruise to take-off power and included data for several

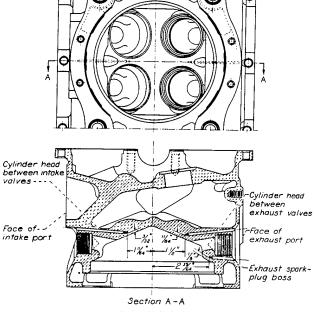


FIGURE 1.-Installation of cylinder-head thermocouples.

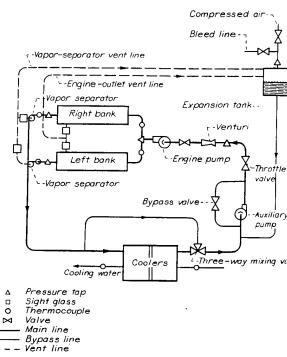


FIGURE 2.-Schematic diagram of engine coolant system.

coolant flows and temperatures and a range of engine cool: outlet pressures from 10 to 30 pounds per square inch ga The effects of coolant temperature and engine power on cylinder temperatures were further determined in the par the investigation conducted on engine B. Tests in wh an aftercooler was mounted on this engine were also made determine the effects of varying the charge flow, the manit temperature, the fuel-air ratio, and the aftercooling cor tions on engine cooling. The part of the investigation engines C and D was conducted to extend the range the cylinder-head-temperature correlation and to prov data essential to both correlations that were not obtained the other two engines.

# ANALYSIS

SYMBOLS The following symbols are used in this analysis:  $\overline{A}$ mean area of cylinder wall, (sq ft) .  $B_{10}$  constants  $B_1$ specific heat of coolant, (Btu)/(lb) (°F) specific heat of air at constant pressure, (Bt (lb) (°F) accleration due to gravity, 32.2 (ft)/(sec<sup>2</sup>) heat rejected to coolant, (Btu)/(sec) IIJ mechanical equivalent of heat, 778 (ft-lb)/(B thermal conductivity of coolant, (Btu)/(s (sq ft) (°F/ft) thermal conductivity of cylinder wall, (Bt (see) (sq ft) ( ${}^{\circ}F/ft$ )

m, n, s

exponents

Α.Υ	engine speed, (rpm)
Pr	Prandtl number of coolant, $c\mu/k$ (dimensionless)
$T_{\epsilon}$	carburetor inlet-air temperature, (°F)
	effective cylinder-gas temperature, (°F)
$T_{\scriptscriptstyle h}$	average cylinder-head (gas side) temperature, (°F)
$T_{h, l}$	average cylinder-head (liquid side) temperature, (°F)
$T_{\iota}$	average coolant temperature, (°F)
$T_m$	dry inlet-manifold temperature, (°F)
$T_m$	supercharger impeller-tip speed, (ft) (sec)
$W_{c}$	engine charge (air plus fuel) flow, (lb)/(sec)
$W_{i}$	coolant flow, (lb)/(sec)
$r_w$	mean thickness of cylinder wall, (ft)
Z	$B_1 \frac{x_w}{k_w A}$ , factor that accounts for temperature
	drop through cylinder head
μ	absolute viscosity of coolant, (lb)/(ft)(sec)

# DERIVATION OF CORRELATION EQUATIONS

Consideration of the process by which heat is transferred from the hot gases in the engine cylinder to the coolant in a liquid-cooled engine indicates that the heat must pass through a series thermal path in the following steps: (1) from the gases in the cylinder to the gas-side cylinder wall; (2) through the cylinder wall; and (3) from the liquid side of the cylinder wall to the coolant. In reference 1, the following three equations were presented for this series thermal path:

Cylinder gases to wall

$$H = B_1 W_c^n (T_g - T_h) \tag{1}$$

where  $T_{\rm g}$  is a function of the fuel-air ratio, inlet-manifold temperature, ignition timing, and exhaust pressure. Through cylinder wall

$$H = \frac{T_h - T_{h, l}}{\frac{x_{to}}{k_{tot} d}} \tag{2}$$

Cylinder wall to coolant

$$H = B_2 \left(\frac{W_l}{\mu}\right)^m \left(\frac{c\mu}{k}\right)^s k \left(T_{h,l} - T_l\right) \tag{3}$$

Cylinder-head-temperature correlation equation.—In order to obtain a single equation expressing the cylinder-head temperature  $T_h$  in terms of the primary engine and coolant variables, equations (1) to (3) are combined in such a manner as to eliminate H and  $T_{h,t}$ . This combination of the three basic equations of heat flow leads to the following cylinder-head-temperature correlation equation:

$$\left[ \left( \frac{T_h - T_l}{T_g - T_h} \right) \left( \frac{1}{W_e^n} \right) - Z \right] k(Pr)^s = B_3 \left( \frac{W_l}{\mu} \right)^{-m} \tag{4}$$

where

$$Z = B_1 \frac{x_w}{k_w A}$$

$$Pr = \frac{c\mu}{k}$$

and

$$B_3 = \frac{B_1}{B_2}$$

In equation (4), the coolant-flow factor  $W_l/\mu$  is separated variable, the effect of charge flow being incorated with the temperature parameter. It may be desir: in many cases, however, to separate the effect of charge fiby rearrangement of equation (4), the following alternation is obtained in which the charge flow is the separatraphe:

$$\left( \frac{T_{\mathsf{g}} - T_{\mathsf{h}}}{T_{\mathsf{h}} - T_{\mathsf{l}}} \right) \left[ \left( \frac{B_3}{W_{\mathsf{l}}^{\mathsf{m}}} \right) \left( \frac{\mu^{\mathsf{m}}}{k \left( P r \right)^{\mathsf{s}}} \right) + Z \right] = W_{\mathsf{c}}^{-\mathsf{n}}$$

The constant  $B_3$ , the factor Z, and the exponents m and s are determined from the test results and the detail their evaluation are given later in this report. The sign cance of other factors appearing in the correlation equal or used in the analysis is discussed in a following sect. When these factors are evaluated, the relations among and the various engine and coolant operating conditions be completely defined by equation (4) or (5), and the equations will then serve to correlate and permit the diction of cylinder-head temperatures for any engine coolant operating condition.

If modes of heat transfer other than normal forced contion are predominant or if variables other than those tained in the correlation equation have an effect on cylinder temperatures, the data may be expected to defrom a satisfactory correlation. Two such factors that the encountered in engine operation are boiling of the cool and scale build-up on the coolant passages. The effect these two factors are illustrated in figures to be present subsequently.

Coolant-heat-rejection correlation equation.—An equal expressing the coolant heat rejection II in terms of primary engine and coolant variables is obtained by a reconstitution of equations (1) to (3) in a manner similar to sused for the derivation of the cylinder-head-temperate correlation equation except that, in this case, the varial  $T_h$  and  $T_{h,I}$  are eliminated. This recombination of the bequations of heat flow leads to the following heat-reject correlation equation:

$$\left[B_1\left(\frac{T_g-T_l}{II}\right)-\left(\frac{1}{W_e^n}\right)-Z\right]k(Pr)^s=B_3\left(\frac{W_l}{\mu}\right)^{-m}$$

As for cylinder-head temperature correlation equation the coolant-flow factor  $W_l/\mu$  is the separated variable in equation and the effect of charge flow is incorporated variable to place the charge flow as the separated variable g the following alternative form for the heat-rejection corr tion equation:

$$\left[B_1\left(\frac{T_s-T_t}{II}\right)-\left(\frac{B_3}{W_t^m}\right)\left(\frac{\mu^m}{k(Pr)^s}\right)-Z\right]=W_c^{-n}$$

The values of the various constants appearing in equation (6) or (7) are evaluated from the data in a manner similar to that used for the cylinder-head-temperature correlation, as illustrated and subsequently described. When these factors are evaluated, the relation between H and the various operating conditions will be completely defined by equation (6) or (7) and these equations will then serve to correlate and permit the prediction of coolant heat rejections for any engine and coolant operating condition.

# SIGNIFICANCE OF FACTORS

Coolant heat rejection H.—The total coolant heat rejection, including the heat rejected from both the cylinder heads and the cylinder barrels, is used in the heat-rejection correlation presented herein. This value of coolant heat rejection was determined from the measured flow and temperature rise of the coolant cooling water and was corrected for an estimated piping loss of 2 percent of the heat rejected.

Cylinder temperature  $T_h$ .—The temperature  $T_h$  used in the correlation of cylinder-head temperatures is the average of the temperatures measured between the exhaust valves for the 12 cylinders of the engine. This temperature location was chosen because it is the most widely used temperature for this engine and, being in the hottest region of the cylinder head, may be considered indicative of critical cooling conditions. Although an average cylinder-head temperature is indicated in the derivation of equations (4) and (5) and is used in the correlation presented in reference 1, a satisfactory correlation would be expected for any single temperature location. This possibility of satisfactorily correlating the temperature of any location is a result of the linear relation (reference 3) that exists between temperatures at various locations in the cylinder head.

In order to permit an evaluation of the maximum cylinderhead temperature obtained for any operating condition, the relation between the average temperature for the 12 cylinders and the temperature of the hottest cylinder is presented.

Effective cylinder-gas temperature  $T_{\rm g}$ .—The effective cylinder-gas temperature  $T_{\rm g}$  is the gas temperature effective in transferring heat from the cylinder gases to the cylinders, and, as previously indicated, is considered a function of the fuel-air ratio, inlet-manifold temperature, ignition timing, and exhaust pressure. The value of  $T_{\rm g}$  for the various engine conditions for the correlation of both cylinder-head temperatures and coolant heat rejections is determined from tests subsequently described and illustrated.

Dry inlet-manifold temperature  $T_m$ .—The true inlet-manifold temperature in a conventional multicylinder engine is difficult to measure because of the presence of unevaporated fuel in the charge mixture. For cooling correlations, the expedient of using a calculated dry inlet-manifold temperature instead of the measured manifold temperature is adopted. This dry inlet-manifold temperature is defined as the sum of the air temperature at the carburetor inlet and the calculated temperature rise of the air incurred in passing through the supercharger. This temperature rise

was calculated on the assumption that there was no vaporization. By assuming a value of 0.96 for the factor, which is the ratio of the pressure coefficient (adiabatic efficiency of the supercharger, the dry manifold temperature  $T_m$  may be written as

$$T_m = T_c + \frac{0.96 U^2}{J_{qC_n}}$$

For the single-stage engines used, this relation reduc-

$$T_m = T_c + 25.28 \left(\frac{N}{1000}\right)^2$$

For the tests of the engine fitted with the aftercoole temperature drop of the charge mixture incurred in parthrough the aftercooler was calculated from the heat rej to the aftercooler coolant and subtracted from the matemperature determined from equation (9).

Charge flow  $W_c$  and fuel-air ratio.—The value of c flow  $W_c$  was taken as the total charge flow (air plus fu the engine, although similar correlations may also be on the basis of the air flow alone. The fuel-air ratio was the mean fuel-air ratio to all cylinders as obtained the total air and fuel flows.

Coolant flow  $W_t$  and coolant temperature  $T_t$ .—The coflow  $W_t$  was, for simplicity, taken as the total coolant to both cylinder banks. Although the construction of coolant passages in the engine is such that the flow of considerably from cylinder to cylinder, it is shown in a ence 3 that changes in the total flow effect proport changes in the flow over any one cylinder. The cotemperature  $T_t$  was taken as the average of the inlet outlet temperatures of both cylinder banks.

Physical properties of coolants.—The physical proper of the coolants (specific heat c, absolute viscosity  $\mu$ , the conductivity k, and therefore the Prandtl number Pr) evaluated at the average coolant temperature  $T_t$ . The ues used were obtained from reference 4 and are prese in convenient curve form in reference 1.

Constants  $B_1$ ,  $B_3$ , and Z and exponents m, n, and so constants  $B_1$ ,  $B_3$ , and Z and the exponents m, n, and determined from the test results, and the details of the unation of these factors are given in the following sec. The values obtained for these constants and exponents, in addition, the value of  $T_g$ , will not necessarily be the for both the cylinder-head-temperature and the heat-reion correlations because in the cylinder-head-temperature the heat-rejection correlation both the cylinder head and cylinder barrel are involved in the transfer of heat to coolant.

# CYLINDER-HEAD-TEMPERATURE CORRELATION EVALUATION OF FACTORS

As previously indicated, the value of the effective temperature  $T_{\mathbf{z}}$  at various engine operating conditions

the values of the constant Z and the exponents m, n, and s must be evaluated before the correlation equations may be used to determine the cylinder-head temperature  $T_h$  for various engine operating conditions. In general, the values of these factors are determined independently from analysis of test data that are selected to permit a simplification of correlation equation (4) as required for this evaluation. The details of the evaluation of these factors are described and illustrated in the following paragraphs.

Effective cylinder-gas temperature  $T_g$ .—The method used to evaluate  $T_s$ , which has been successfully applied to the correlation of cooling data obtained for a large number of air-cooled engines and the liquid-cooled engine of reference 1, constitutes first the establishment of a reference value of  $T_{\rm g}$  for a given set of operating conditions and then the determination of the variation of  $T_{\varepsilon}$  with each of the pertinent engine conditions. On the basis of previous correlation work, and in the interest of consistency with other cylinder-head-temperature correlations (references 1, 5, 6, and 7), a reference value of  $T_s$  for this correlation of 1150° F was chosen for a fuel-air ratio of 0.080, an inlet-manifold temperature of 80° F, standard ignition timing (approximately maximum power setting), and an exhaust pressure of 30 inches of mercury absolute. Investigation has shown that the correlation is insensitive to changes in the magnitude of this reference value of  $T_{g}$ ; it is important, however, that its variation with engine conditions be accurately determined.

The variation of  $T_s$  with fuel-air ratio, manifold temperature, ignition timing, and exhaust pressure was determined from the tests in which these factors were independently varied while holding all other engine and coolant conditions constant. For such conditions, correlation equation (4) reduces to

$$\frac{T_h - T_l}{T_s - T_h} = \text{constant}$$
 (10)

This constant can be evaluated from the cylinder-head and coolant-temperature data obtained at the reference operating condition for which the value of  $T_{\kappa}$  has already been chosen. The variation of  $T_{\kappa}$  with each of the aforementioned variables can then be calculated from the values of the constant and the coolant and cylinder-head temperatures obtained for the range of operating conditions in question. Although the value of the constant is dependent on the charge flow and coolant conditions of the reference operating condition and thus was not the same for each series of runs for which the variation of  $T_{\kappa}$  was being established (see table I), the variation of  $T_{\kappa}$  from the chosen reference value is independent of the value of the constant and hence is valid for any engine charge flow and coolant operating condition.

The variation of  $T_s$  with fuel-air ratio is shown in figure 3. The values of  $T_s$  presented have been corrected to a dry inlet-manifold temperature of 80° F in accordance with a

relation between the manifold temperature and  $T_{\rm x}$  that we be subsequently discussed. A maximum value of  $T_{\rm x}$  reached at a fuel-air ratio of about 0.067, which is appromately equal to the fuel-air ratio for the stoichiometric meture. Data obtained from three different engines, one which was fitted with an aftercooler, are included in figure and close agreement among the three is noted. The varitions obtained for both the single-cylinder engine of reference 1 and the air-cooled engine of reference 5 are also show in this figure. The values of  $T_{\rm x}$  for the single-cylindering are seen to be somewhat lower than those for the multicylinder engines of the present investigation, part ularly in the rich region, but close agreement between the multicylinder liquid-cooled and air-cooled engines is eviden

The variation of  $T_{\kappa}$  with the calculated dry inlet-manifor temperature  $T_m$  is shown in figure 4. The data, which a presented for three engines and various engine operation conditions, have been adjusted to a fuel-air ratio of 0.080 accordance with the relation between  $T_{\kappa}$  and the fuel-aratio that is presented in figure 3. As indicated in equation the inlet-manifold temperature may be varied by chaning either the carburetor inlet-air temperature or the engispeed. Data obtained from tests wherein each of the quantities was independently varied are presented (fig. and both sets of data fall on a common curve. An average

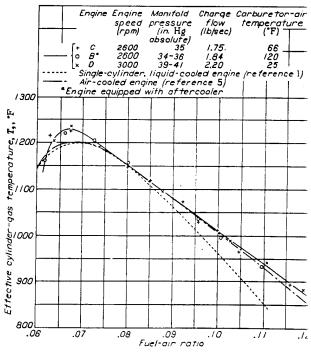


FIGURE 3.—Variation of effective cylinder-gas temperature with fuel-air ratio for cylind head-temperature correlation. Data corrected to dry inlet-manifold temperature of 80° exhaust pressure, 29-30 inches mercury absolute; standard ignition timing.

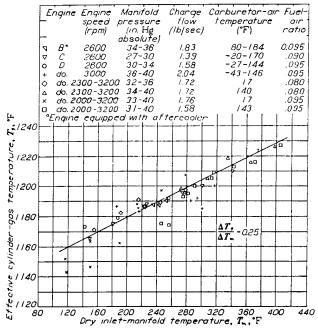


FIGURE 4.—Variation of effective cylinder-gas temperature with manifold temperature for cylinder-head-temperature correlation. Data corrected to fuel-air ratio of 0.080; exhaust pressure, 29-30 inches mercury absolute; standard ignition timing.

change in  $T_s$  of about 0.25° F per degree Fahrenheit change in  $T_m$  is indicated and correction of  $T_s$  to other than 80° F manifold temperature is therefore made in accordance with the relation

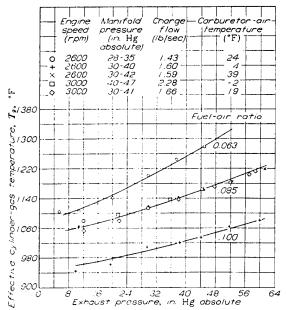


Figure 5.—Variation of effective cylinder-gas temperature with exhaust pressure at several fuel-air ratios for cylinder-head-temperature correlation. Data corrected to dry inlet-manifold temperature of 80° F; standard ignition timing; engine D.

$$\Delta T_{\rm g} = 0.25 \ (T_{\rm m} - 80)$$

The effect of exhaust pressure on  $T_{\kappa}$  for three valued-air ratio and various engine conditions is presentigure 5. These values of  $T_{\kappa}$  have been corrected inlet-manifold temperature of 80° F by means of equation An increase in exhaust pressure results in an increase and, for the range of fuel-air ratios covered, the in is somewhat greater at the lean than at the rich tures. A similar effect of exhaust pressure on  $T_{\kappa}$  with the tests of an air-cooled engine, which are rejin reference 6, and also in tests of another liquid-engine conducted at this laboratory (data unavailable), convenience, a cross plot of these curves is shown in fining which  $T_{\kappa}$  is plotted as a function of fuel-air rate exhaust pressures of 10, 20, 30, 40, and 50 inches of meabsolute. The curve for an exhaust pressure of 30

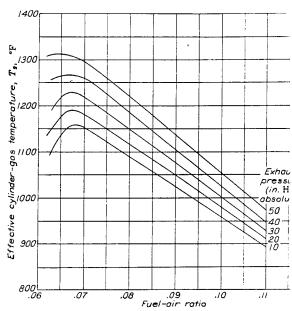


FIGURE 6.—Variation of effective cylinder-gas temperature with fuel-air ratio at exhaust pressures for cylinder-head-temperature correlation (obtained from crosfig. 5). Data corrected to dry inlet-manifold temperature of 80° F: standard timber grains D.

of mercury was obtained from figure 3 and the shape curve was used as a guide in drawing those for the exhaust pressures.

The variation of  $T_s$  with ignition timing for engines of 2600 and 3000 rpm is presented in figure 7 as a p  $\Delta T_s$  against ignition timing. This variable  $\Delta T_s$  reprethe change in effective cylinder-gas temperature from value at the standard ignition timing (exhaust spark) of 34° B.T.C. The magnitude of the correction to other than standard ignition timing increases positive

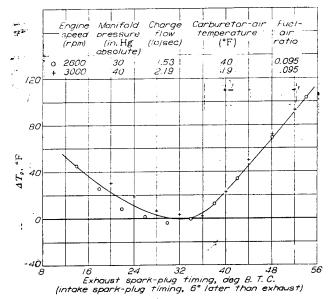


FIGURE 7.—Variation of change in effective cylinder-gas temperature with ignition timing for cylinder-head-temperature correlation. Engine D.

the spark setting is increased or decreased from the norma position. The data separate somewhat with engine speed for a spark-plug timing later than the normal setting but for simplicity, a single curve has been drawn through all the data.

Exponent n on charge flow.—The value of the exponent n on charge flow  $W_c$  was obtained from tests at constant coolant conditions in which the charge flow was varied by changing the engine speed, manifold pressure, or exhaust pressure. For such conditions, equation (4) reduces to

$$\frac{T_h - T_l}{T_k - T_h} = B_4 W_c^n \tag{12}$$

A logarithmic plot of  $T_h - T_t$  against  $W_c$  is shown in figure 8 and the slope of the line through the data, which is equal to the value of the exponent n, is 0.60. Although the absolute value of the factor  $T_k - T_t$  would be different for different coolant conditions, the slopes of the resulting lines would be the same. The data for the runs with a coolant flow of 300 gallons per minute were adjusted to a flow of 250 gallons per minute to be consistent with the rest

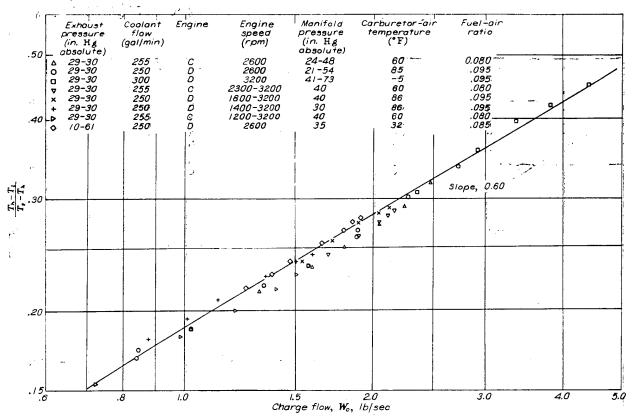


Figure 8.—Determination of exponent n on charge flow  $W_{\epsilon}$  for cylinder-head-temperature correlation from variation of  $\frac{T_h - T_l}{T_{\epsilon} - T_h}$  with  $W_{\epsilon}$ . Coolant, 30-70 ethylene glycol—water; average coolant temperature, 245° F.

of the data by changing the factor  $\frac{T_h - T_I}{T_s - T_h}$  in accordance with the effect of coolant flow on the cylinder-head temperature presented in reference 3. Data covering a wide range of engine speed, manifold pressure, and exhaust pressure are presented for two engines and close agreement is noted.

Factor Z.—The factor Z was determined from tests in which the coolant temperature and composition were held constant while the coolant flow  $W_I$  was varied. For these conditions, equation (4) may be written

$$\left( \frac{T_h - T_l}{T_s - T_h} \right) \left( \frac{1}{W_c} \right) - Z = B_5 \left( \frac{1}{W_l} \right)^m$$
 (13)

A plot of  $\binom{T_h-T_l}{T_\kappa-T_h}\binom{1}{W_c^{0.50}}$  against  $1/W_l$ , using the previously determined values of  $T_\kappa$  and the exponent n, is shown in figure 9 for five different coolant compositions. Extrapolation of these data to  $1/W_l=0$  (at which point the value of  $\binom{T_k-T_l}{T_\kappa-T_h}\binom{1}{W_c^{0.50}}$ ) is equal to the Z factor) results in a value of 0.13 for Z. Because of the somewhat arbitrary nature of this extrapolation, the final value for Z was chosen by drawing curves having the same general shape as those presented for a similar plot in reference 1 and then completing and plotting a final correlation of all the data (based on equation (4)) using three different values for Z in the neighborhood of the value indicated by the curves; the value that gave the most satisfactory correlation (0.13) was finally chosen.

In the investigation of reference 3, it was found that the cylinder-head temperature, particularly in the exhaust or hot side of the head, increased with engine running time during the initial operation of the engine. This increase in temperature is illustrated in figure 10 and it is noted that the temperature between the exhaust valves increased about 25° F during approximately the initial 100 hours of engine running time and remained substantially constant as the operating time was further increased. Unlike the variation exhibited by this temperature location, the temperature at

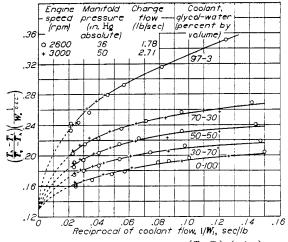


Figure 9.—Determination of factor Z from variation of  $\left(\frac{T_h - T_l}{T_e - T_h}\right) \left(\frac{1}{W_e^{0.60}}\right)$  with  $1/W_l$  for cylinder-head-temperature correlation. Average coolant temperature, 245° F; engine D.

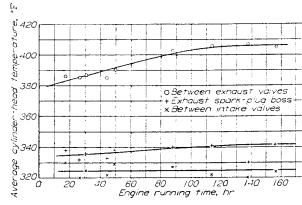


FIGURE 10.—Variation of average cylinder-head temperatures with engine running Engine speed, 2020 rpm; manifold pressure, 32 inches mercury absolute; charge flow pounds per second; Incl-air ratio, 0.095; carburetor-air temperature, 82° F; coolant, ethylene glycol—water; coolant flow, 300 gallons per minute; average coolant temperature, 43° F; standard ignition timing; exhaust pressure, 29-30 inches mercury absolute; engineer and perfectly approximately approxi

the exhaust spark-plug boss increased only slightly and temperature between the intake valves remained const over the entire period of the investigation. As discussed reference 3, an inspection of the coolant passages of a scrap cylinder head revealed scale deposits on the exhaust side the cylinder head but none on the intake side. This incre in temperature was therefore attributed to the scale depo on the coolant passages. Because the Z factor accounts the temperature drop through the cylinder head, this incre in cylinder-head temperature with engine running time be reflected as a similar variation in the Z factor. The vaof the Z factor (0.13), which was determined after about hours accumulated engine running time, is therefore ap cable only for this or greater engine running times. Altho the final correlation is insensitive to the magnitude of basic value of Z, it is necessary that the variation of Z, wh occurs during the initial period of engine running time. accurately determined and included in the final correlat

The variation of Z during the initial period of enrunning time was determined from cylinder-head-temperatdata obtained at a reference operating condition. For reference condition, the coolant conditions were constraccordingly, equation (13) may be written

$$\left(\frac{T_h - T_l}{T_v - T_h}\right) \left(\frac{1}{W_0^{0.60}}\right) - Z = \text{constant}$$

The value of the constant is determined from the substitution to the equation of the previously determined value of Z, the pertinent engine and coolant data obtained at 100 herogine running time. The value of Z at other engine runnitimes is then determined from the value of the constand the cylinder-head temperatures obtained at the runnitime under consideration.

A plot showing the variation of the Z factor for engine with engine running time is presented in figure 11. It can seen that the Z factor increases during the initial engoperating time in a manner similar to that for the cylinc head temperature (fig. 10) and that there is no signific change in the value of Z after about 100 hours running to It is expected that this phenomenon will vary from engine engine depending upon the history of operation. The rat

of conditions and engine running time over which this variation was established is indicated in table I.

For engine D, the value of Z used in the subsequent cylinder-head-temperature correlation plots was determined from the curve of figure 11. For engines A, B, and C, the data were insufficient to provide separate evaluation of Z, but most of the correlation data provided by these engines were obtained after the engines had been run for a considerable length of time so a value of 0.13 was used. Because this value resulted in satisfactory correlation of the cylinder-head-temperature data from engines A, B, and C with those of engine D, it may be assumed that the coolant passages of these engines were in about the same condition as those of engine D after about 100 hours engine running time.

Exponent m on coolant-flow parameter  $W_t/\mu$ .—The value of the exponent m on the coolant-flow parameter  $W_t/\mu$  was determined from tests in which the coolant temperature and

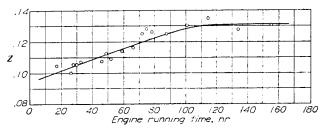


FIGURE 11.—Variation of factor Z for cylinder-head-temperature correlation with initial engine running time attributed to scale build-up in coolant passages of engine D.

composition were held constant while the coolant flow  $W_I$  was varied (similar to data for determination of Z factor). For these conditions, equation (4) may be written

$$\left[ \left( \frac{T_h - T_l}{T_g - T_h} \right) \left( \frac{1}{W_c^{\bar{n}}} \right) - Z \right] k = B_h \left( \frac{W_l}{\mu} \right)^{-m} \tag{15}$$

A logarithmic plot of the factor 
$$\left[\binom{T_h-T_i}{T_g-T_h}\binom{1}{W_e^{0.00}}-Z\right]k$$

against  $W_l/\mu$  is shown in figure 12 for five different coolant compositions. The slope of the straight lines through these data is equal to the exponent on  $W_l/\mu$ . Lines having the same slope are drawn through the data for each coolant and the value of the exponent m is accordingly established as 0.48. This value for the exponent m, which is established from these selected data, will, of course, be verified by the slope of the line through all the data on the final correlation plot based on equation (4).

Exponent s on Prandtl number Pr.—The value of the exponent s on the Prandtl number Pr was determined from data for a constant value of  $W_t/\mu$ ; for this condition, equation (4) may be reduced to

$$\left[ \left( \frac{T_h - T_t}{T_g - T_h} \right) \left( \frac{1}{W_e^n} \right) - Z \right] k = B_7 (Pr)^{-s}$$
 (16)

The slope of the line determined by a logarithmic plot of  $\left[\binom{T_h-T_l}{T_s-T_h}\binom{1}{W_c^{0.60}}-Z\right]k \text{ against } Pr \text{ would then establish the value of the exponent } s. A wide range of Prandtl number for this plot may be obtained if data for several different$ 

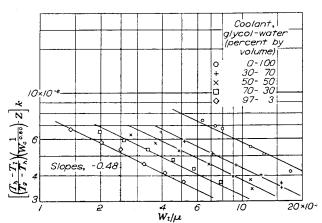


Figure 12.—Determination of exponent m on coolant-flow parameter  $Wd\mu$  for cylinder-heat temperature correlation from variation of  $\left[\left(\frac{T_k-T_l}{T_e-T_k}\right)\left(\frac{1}{W_e^{0.06}}\right)-Z\right]k$  with  $Wd\mu$  k various coolants. Average coolant temperature, 245° F; engine D.

coolants are used. In order to construct this plot, it is convenient to obtain values of the factor

$$\left[ \left( \frac{T_{\scriptscriptstyle h} - T_{\scriptscriptstyle l}}{T_{\scriptscriptstyle g} - T_{\scriptscriptstyle h}} \right) \left( \frac{1}{W_{\scriptscriptstyle 0}^{0.60}} \right) - Z \right] k$$

from figure 12 for a constant value of  $W_l/\mu$  and then to cross-plot the values of this factor against the Prandtl number Pr of the different coolants. Data obtained from a crosplot of figure 12 at a constant value of  $W_l/\mu$  equal to 55,000 is shown in figure 13 and the slope of this line thus established the value of the exponent s on the Prandtl number Pr as 0.33

# FINAL CORRELATION

Final correlation with coolant-flow factor  $W_l/\mu$  as in dependent variable.—The final correlation based on equation (4), which is obtained by plotting the factor

$$\left[ \binom{T_h - T_l}{T_e - T_h} \binom{1}{W_c^{0.60}} - Z \right] k \ (Pr)^{0.33}$$

against  $W_l/\mu$  on logarithmic coordinates for all test data for the four engines, is presented in figure 14(a). Although the data points scatter considerably, the maximum variation

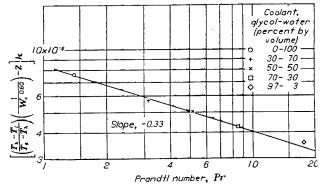
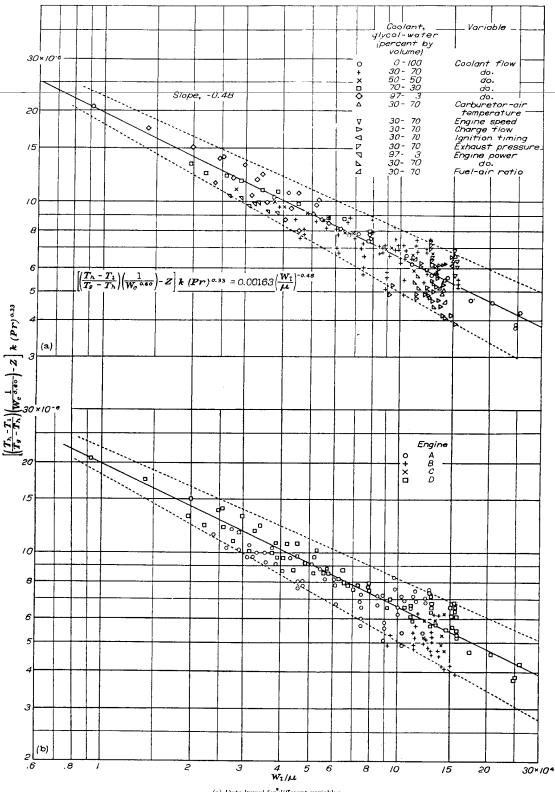


FIGURE 13.—Determination of exponents on Prandil number Pr for cylinder-head-temperature correlation from variation of  $\left[\left(\frac{T_h-T_l}{T_s-T_h}\right)\left(\frac{1}{W_c^{0.50}}\right)-Z\right]k$  with Pr as obtained from cross plot of figure 12 at value for  $W_l/\mu$  of 55,000. Average coolant temperature, 245° F country 11.



(a) Data keyed for different variables,

(b) Data keyed for different engines.

 ${\tt Figure~14.-Final~correlation~of~cylinder-head~temperatures~based~on~equation~(4).}$ 

in  $T_{\hbar}$  resulting from this scatter is not large. A variation in  $T_{\hbar}$  of  $\pm 12^{\circ}$  F for typical engine conditions at normal rated power is represented by the dashed lines in the figure, and it is seen that almost all the data lie within this band.

The value of the exponent m, which is obtained from the slope of the line through the data, is, as previously determined, equal to 0.48 and the value of the constant  $B_3$ , found by substitution of the values of the coordinate of any point on the line into the correlation equation, is equal to 0.00163. The final equation is accordingly written

$$\left[ \left( \frac{T_h - T_t}{T_g - T_h} \right) \left( \frac{1}{W_e^{0.60}} \right) - Z \right] k \left( Pr \right)^{0.33} = 0.00163 \left( \frac{W_t}{\mu} \right)^{-0.48} \tag{17}$$

and will apply over a considerable range of engine operating conditions, coolant temperatures, coolant flows, and coolant compositions. In order to illustrate the variation in the correlation of tained among the four engines, the correlation of figure 14 (a is replotted in figure 14(b), using a different symbol fo each engine. The separation of the data between engine is slight and the scatter for one engine is almost as great a the total scatter for all engines.

Final correlation with charge flow  $W_c$  as independen variable.—Correlation of the test data based on equation (5)

wherein the factor 
$$\binom{T_s-T_h}{T_h-T_t}\left[\binom{0.00163}{W_t^{0.48}}\right)\left(\frac{\mu^{0.48}}{k(Pr)^{0.33}}\right)+Z\right]$$

is evaluated by using the previously determined values of m, s, Z, and  $B_3$  and plotted against the charge flow  $W_c$ , is shown in figure 15. A straight line with a slope of -0.60 which is equal in absolute value to the value of the exponen n on  $W_{c_1}$  previously determined, is drawn through the

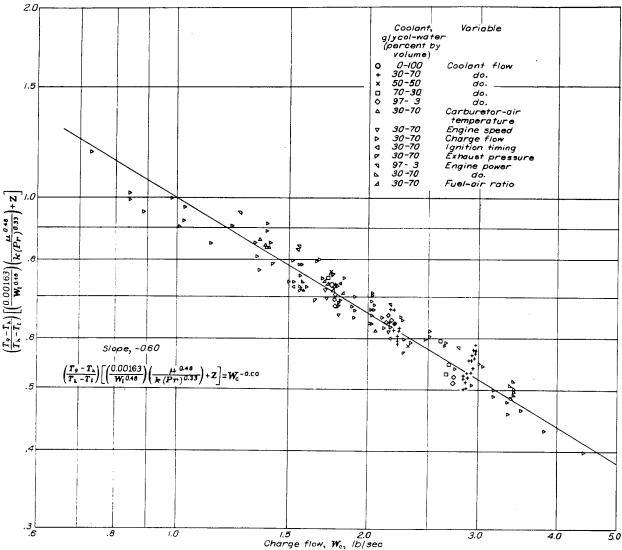


FIGURE 15.—Final correlation of cylinder-head temperatures based on equation (5)

data. As a result of the different arrangement of the terms in this equation, the scatter of the data is considerably less in figure 15 than in figure 14; the over-all accuracy of the correlation is, of course, the same.

When equation (17) is rearranged in accordance with this plot, the following form is obtained:

$$\begin{pmatrix} T_s - T_h \\ T_h - T_t \end{pmatrix} \left[ \begin{pmatrix} 0.00163 \\ W_t^{0.48} \end{pmatrix} \begin{pmatrix} \frac{\mu^{0.48}}{k(Pr)^{0.33}} \end{pmatrix} + Z \right] = W_c^{-0.60}$$
 (18)

As previously mentioned, the values of the factor Z for engine D used in plotting the final correlation of figures 14 and 15 were obtained from figure 11. If a constant average value were used for Z, the scatter of the data would be increased from approximately  $\pm 12^{\circ}$  to about  $\pm 22^{\circ}$  F, depending upon the engine running time.

In order to facilitate the computation of head temperatures by means of equation (18), values of the coolant-property parameter  $\binom{\mu^{0.18}}{k(Pr)^{0.33}}$  are presented in figure 16 for various coolant mixtures of ethylene glycol and water over a range of coolant temperatures.

Effect of boiling of coolant on correlation.—It was found in the investigation of reference 3 that under some conditions of operation a reduction in coolant flow increased the amount of boiling of the coolant thus reducing the normal tendency of the cylinder-head temperature to increase with reduced coolant flow. In order to determine the effect of this boiling of the coolant on the head-temperature correlation, the rela-

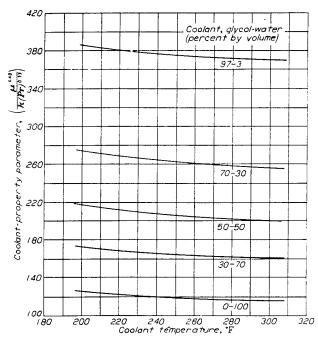


Figure 16.—Variation of coolant-property parameter  $\left(\frac{\mu^{0.48}}{k(Pr)^{0.33}}\right)$  with coolant temperature for cylinder-head-temperature correlation.

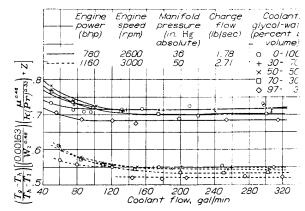


FIGURE 17..-Variation of  $\left(\frac{T_k - T_h}{T_h - T_f}\right) \left[\left(\frac{0.00163}{W_0^2 0.48}\right) \left(\frac{\mu^{0.48}}{k (Pr)^{0.33}}\right) + Z\right]$  with coola as indication of boiling. Average coolant temperature, 245° F; engine D.

tions given by equation (18) were considered. The hand side of this equation should normally be a function only charge flow and, if boiling of the coolant were negligwould be independent of the coolant flow. If, how boiling occurs to an appreciable extent, the value of left-hand side of this equation would be expected to with the coolant flow. A plot of this parameter ag coolant flow is shown in figure 17 for two different er operating conditions and several coolant compositions. coolant flows greater than about 100 gallons per minute value of the plotted parameter is independent of the coflow, which indicates that boiling of the coolant did not e to a noticeable degree in this range. For coolant flowthan about 100 gallons per minute, however, the value o parameter increases with reduced coolant flow (deper on the engine power and coolant), which illustrates the t ency of boiling of the coolant in this range of flow rat reduce the cylinder-head temperature.

The maximum variation of the parameter plotted in figures equivalent to a decrease in head temperature of than 10° F. Because this variation is within the no scatter of the data, the correlation is not seriously affeby boiling for the ranges of variables covered. Extrapole of this correlation to combinations of higher engine pelower coolant flows, or lower coolant pressures than a covered by this investigation would, however, be subjective transfer of the cooleant pressures and reduced accuracy of prediction of cylindrical contents.

Relation between maximum and average cylinder-temperature.—In figure 18, the cylinder-head temperature of the exhaust valves of the hottest cylinder is pleagainst the average temperature of the 12 cylinders at location. Good correlation of these data is obtained for conditions and for all engines with the maximum cylinhead temperature ranging from 10° to 20° F higher than average temperature. From the correlation of the average temperature with the primary engine variand coolant variables given in figure 14 or 15 and from

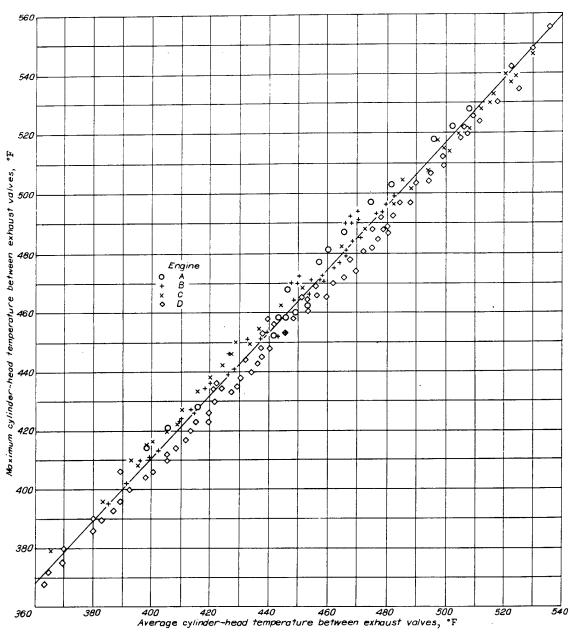


FIGURE 18.—Relation of average cylinder-head temperature between exhaust valves for 12 cylinders to hottest cylinder temperature measured at same location.

relation between the maximum and average head temperatures shown in figure 18, an estimation of the maximum cylinder-head temperature between exhaust valves is possible for a large range of engine and coolant conditions.

# COOLANT-HEAT-REJECTION CORRELATION EVALUATION OF FACTORS

As previously discussed, the values of the various constants and exponents appearing in the correlation equations may

not necessarily be the same for both the cylinder-heatemperature correlation and the coolant-heat-rejection celation because only the heat-transfer processes occurring part of the cylinder head are involved in the cylindhead-temperature correlation, whereas the heat-transprocesses occurring in the complete engine cylinder involved in the heat-rejection correlation. The method evaluating the various constants and exponents for a coolant-heat-rejection correlation is, however, similar to the

previously discussed for the cylinder-head-temperature correlation and the details of their determination are given in the following paragraphs.

Effective cylinder-gas temperature  $T_{\rm g,c}$ -Because of differences in the heat-transfer processes affecting the cylinder-head temperatures and the coolant heat rejections just discussed, the reference value of  $T_{\rm g}$  of 1150° F that was used for the cylinder-head-temperature correlation was considered insuitable for application to a correlation of coolant-heat-rejection data. Accordingly, a reference value of  $T_{\rm g}$  for the coolant-heat-rejection correlation was determined from consideration of equation (1)

$$H = B_1 W_c^n (T_g - T_h) \tag{1}$$

The relation between H and  $T_h$  for constant  $W_c$  and  $T_g$  is obtained from runs in which one of the coolant variables is varied while all the engine conditions are held constant. These values of H are then plotted against  $T_h$  and extrapolated to zero H; the value of  $T_h$  at zero H will then be equal to the value of  $T_g$  at the particular engine operating condition.

The value of  $T_h$  indicated in equation (1) should be an average inside wall temperature for the entire cylinder, but inasmuch as the instrumentation was insufficient to provide an average wall temperature, the average of the temperatures measured between the exhaust valves of the 12 cylinders was used. Although the resulting reference value for  $T_g$  is somewhat higher than what would have been obtained if an average cylinder temperature were used, this procedure is considered satisfactory; as previously mentioned for the cylinder-temperature correlation, the accuracy of the final correlation depends primarily on the accuracy of the final correlation of  $T_g$  with the various engine operating conditions and is insensitive to fairly large changes in its reference value.

The resulting plot of H against  $T_n$  obtained from runs in which the coolant temperature  $T_t$  was varied for two different coolants is shown in figure 19. Extrapolation of the curves to zero H, at which point the average head temperature is equal to the effective gas temperature, gives an initial value for  $T_{\pi}$  of about 750° F for a fuel-air ratio of 0.095, a dry inlet-manifold temperature of 254° F, an exhaust pressure of 29 to 30 inches of mercury absolute, and standard ignition timing. When this initial value of  $T_{\kappa}$  is corrected to the customarily assumed reference conditions of fuel-air ratio of 0.080, dry inlet-manifold temperature of 80° F, exhaust pressure of 29 to 30 inches of mercury absolute, and standard ignition timing in accordance with the relations to be presented later, a value of 760° F is obtained. Although this reference value of  $T_{\rm g}$  for the coolant-heatrejection correlation is considerably lower than the value of 1150° F used in the cylinder-head-temperature correla-

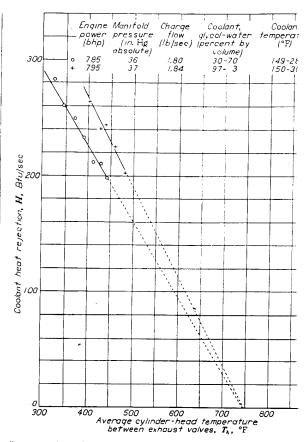


FIGURE 19.—Determination of  $T_{\theta}$  from variation of coolant heat rejection with eylinder-head temperature between exhaust valves for coolant-heat-rejection corr-Coolant flow, 200 gallons per minute; engine speed, 2600 rpm; fuel-air ratio, 0.095; dramanifold temperature, 254° F; exhaust pressure, 29–30 inches mercury absolute; stignifion timing.

tions and approaches the value of 600° F used in cylinbarrel-temperature correlations (references 1 and 7), considered suitable for the present correlation of cocheat rejections in view of the satisfactory final correlation obtained

The variation of  $T_{\rm g}$  with fuel-air ratio, inlet-man temperature, ignition timing, and exhaust pressure for heat-rejection correlation is determined in a manner sit to that for the head-temperature correlation from the in which these factors were each independently varied  $\nu$  holding all other conditions constant. For these conditionrelation equation (6) becomes

$$\frac{T_{\rm g} - T_{\rm l}}{II} = {\rm constant}$$

This constant is evaluated from the heat-rejection coolant-temperature data at the operating conditions

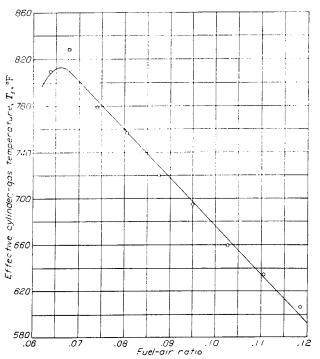


FIGURE 20.—Variation of effective cylinder-gas temperature with fuel-air ratio for coolantheat-rejection correlation. Data corrected to dry inlet-manifold temperature of 80° F; exhaust pressure, 29-30 inches mercury absolute; standard ignition timing.

which the value of  $T_{\rm g}$  has already been established. The variation in  $T_{\rm g}$  with each of the aforementioned variables is then calculated from the value of the constant and the heat-rejection and coolant-temperature data obtained as the operating condition in question.

The variation of  $T_{\rm g}$  with fuel-air ratio is shown in figure 20. The data have been corrected to a dry inlet-manifold temperature of 80°. F in accordance with the relation between  $T_{\rm g}$  and  $T_{\rm m}$  to be presented later. Although the data points do not clearly define a curve in the region of stoichiometric fuel-air ratio, the shape of the curve in this region was made similar to that determined for similar relations in the cylinder-head-temperature correlations.

The variation of  $T_s$  with the calculated dry inlet-manifold temperature  $T_m$  is presented in figure 21. These data, which include both variable carburetor-air-temperature and variable engine-speed runs, have been corrected to a fuel-air ratio of 0.080 in accordance with the relation between  $T_s$  and fuel-air ratio presented in figure 20. Although there is considerable scatter of the data, the trends indicated by both types of run are the same. A line through the average of the data indicates an increase in  $T_s$  of about 0.30° F per degree Fahrenheit increase in  $T_s$ ; correction of  $T_s$  to other than 80° F inlet-manifold temperature is therefore made according to the following relation:

$$\Delta T_{\rm g} = 0.30 (T_{\rm m} - 80) \tag{20}$$

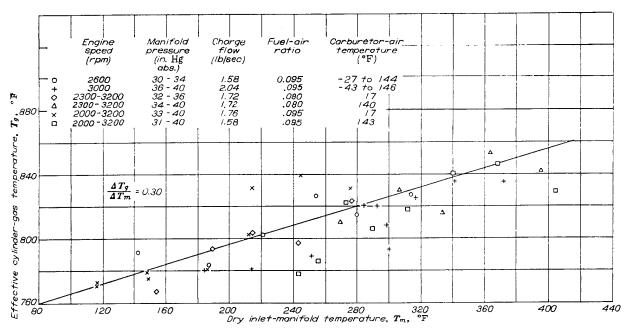


Figure 21.—Variation of effective cylinder-gas temperature with inlet-manifold temperature for coolant-heat-rejection correlation. All data corrected to fuel-air ratio of 0.080; exhaust pressure, 29-30 inches mercury absolute; standard ignition timing.

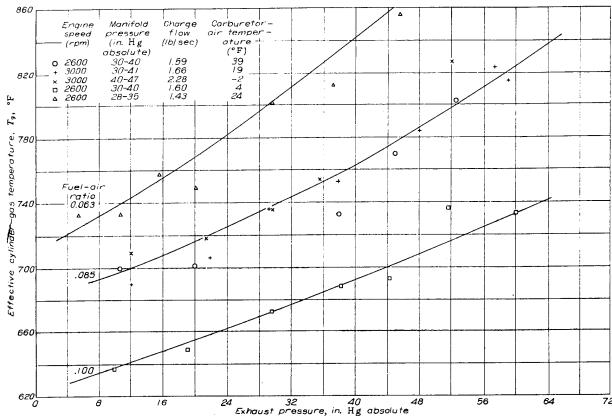


Figure 22. - Variation of effective cylinder-gas temperature with exhaust pressure at several fuel-air ratios for coolant-heat-rejection correlation. Data corrected to dry inlet-manifol perature of 80° F; standard ignition timing.

The effect of exhaust pressure on  $T_{\rm g}$  for three values of fuel-air ratio and a variety of engine conditions is presented in figure 22. All the data have been corrected to an inlet-manifold temperature of 80° F in accordance with equation (20). As for the cylinder-head-temperature correlation, the increase in  $T_{\rm g}$  with increased exhaust pressure is greater at the lean than at the rich mixtures for the range of fuel-air ratios investigated. For convenience, a cross plot of the curves in figure 22 is shown in figure 23 in which  $T_{\rm g}$  is plotted as a function of fuel-air ratio for various exhaust pressures. The curve for an exhaust pressure of 30 inches of mercury was obtained from figure 20 and served as a guide for the fairing of the other curves.

The variation of  $T_{\rm g}$  with ignition timing for engine speeds of 2600 and 3000 rpm is presented in figure 24 as a plot of  $\Delta T_{\rm g}$  against ignition timing. This curve has a minimum at an exhaust spark-plug timing of about 30° B.T.C. and increases as the spark is advanced or retarded from this setting.

Exponent n on charge flow  $W_c$  and constant  $B_1$ .—The value of the exponent n on charge flow  $W_c$  and the constant  $B_1$  were obtained from the series of runs at constant coolant conditions in which the charge flow was varied by changing either the manifold pressure or engine speed. (A summary of the conditions for these runs is given under variable charge

flow in table I.) For such conditions, correlation equation reduces to

$$B_1\left(\frac{T_s-T_l}{II}\right)-\frac{1}{W_c}=\text{constant}$$

According to this equation, a plot of  $1/W_c^n$  ag:  $(T_s - T_t)/H$  on rectangular coordinates would define a straline having a slope equal to the value of  $B_1$  provided the value of n were properly chosen. Because this merequires a trial-and-error solution and may not be as sensias desired, a second method of determining n and  $B_t$  be on equation (1) was used.

According to equation (1), a plot of  $W_c$  against  $H/(T_{\kappa^-})$  on logarithmic coordinates would result in a line having slope equal to n and a value of  $B_1$  determined from coordinates of any point on the line. These values of n  $B_1$  may then be verified for use in equation (21), as previor discussed. Such a plot is shown in figure 25, wherein in values of 0.94 for n and 0.37 for  $B_1$  are obtained.

The plot of  $1/W_c^n$  against  $(T_s - T_l)/H$ , wherein the invalue of 0.94 was substituted for n, is presented in figure A straight line having a slope of 0.37 satisfactorily represthe data and thus the values of 0.94 for n and 0.37 for B verified. Close agreement is seen to exist between the vable inlet-manifold-pressure and variable engine-speed d

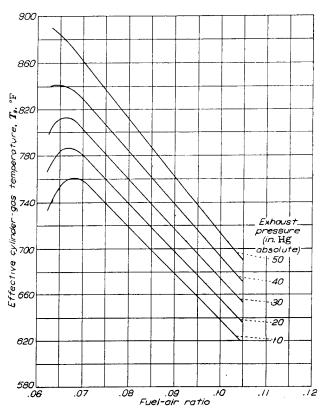


FIGURE 23.—Variation of effective cylinder-gas temperature with fuel-air ratio for various exhaust pressures for coolant-heat-rejection correlation (obtained from cross plot of fig. 22). Data corrected to dry inlet-manifold temperature of 80° F; standard ignition timing.

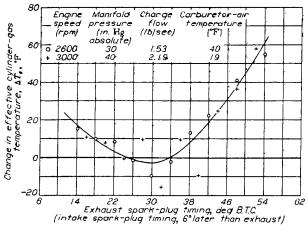


FIGURE 24.—Variation of change in effective cylinder-gas temperature with ignition fiming for coolant-heat-rejection correlation. Fuel-air ratio, 0.095.

Factor Z.—For the operating conditions of constant coolant temperature and composition and variable coolant flow  $W_i$ , which provide data for the determination of the factor Z, equation (6) reduces to

$$B_1\left(\frac{T_s - T_l}{H}\right) - \frac{1}{W_c^n} - Z = B_s\left(\frac{1}{W_l}\right)^m \tag{22}$$

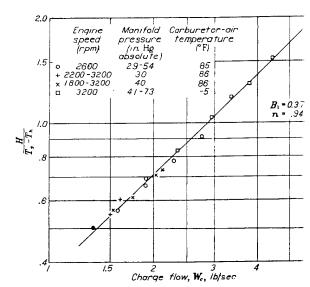


FIGURE 25.—Determination of exponent n and constant  $B_0$ ; or coolant-heat-rejection c-tion from variation of  $H/(T_0 - T_0)$  with  $W_{s_0} = H$ eleair ratio, 0.005; exhaust pressure inches mercury absolute; standard ignition timbus.

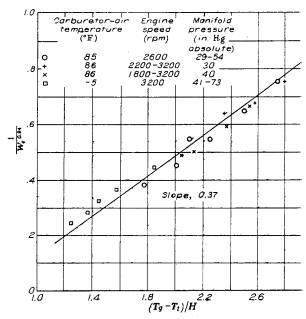
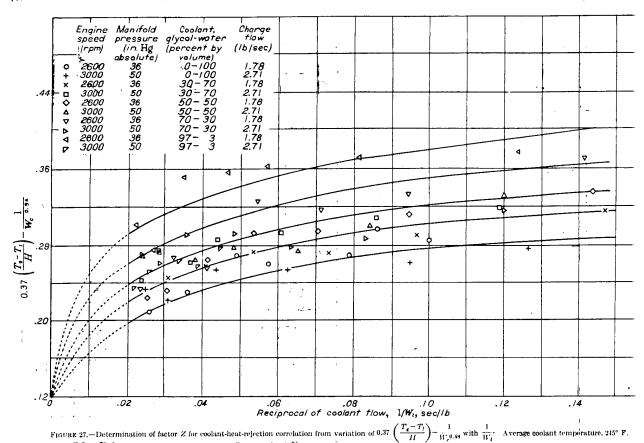


Figure 26.—Variation of  $1/We^{0.9}$  with  $(T_{\theta}-T_{\theta})/H$ . Fuel-air ratio, 0.005; exhaust pre 29-30 inches mercury absolute; standard ignition timing.

Inspection of this equation indicates that  $Z=B_1\left(\frac{T_s-T_t}{H}\right)$   $\frac{1}{W_c^n}$  when  $\frac{1}{W_t}=0$ ; therefore a plot of  $B_1\left(\frac{T_s-T_t}{H}\right)$  against  $\frac{1}{W_t}$  using the previously determined values of  $B_t$ 

and  $T_{\rm g}$  would give an indication of the value of the Z factor when extrapolated to zero  $1/W_L$ . Such a plot is shown figure 27 for five different coolants at two engine powers and value of 0.12 is indicated for Z. As for the cylinder-he temperature correlation, several values in the neighborhood

\*



of 0.12 were chosen for Z and used in trial correlation plots; the value that gave the most satisfactory correlation (0.12) was finally used.

Although it would be expected that the scaling of the coolant passages in the hot regions of the cylinder head (illustrated by the temperature curves of fig. 10) would have an effect on the coolant heat rejection similar to that which it had on the cylinder-head-temperature correlation (fig. 11), no such effect could be detected within the accuracy of the data. It is likely, however, that the effect of this scaling on the heat-rejection correlation is less than those for the head-temperature correlation because the scale was deposited in only a portion of the cylinder head and thus would not have as great an effect on the total coolant heat rejection as it did on the cylinder-head temperature between the exhaust valves.

Exponent m on coolant-flow parameter  $W_1/\mu$ .—For the determination of the exponent m on the coolant-flow parameter  $W_1/\mu$ , data similar to that used for the determination of Z are used and equation (6) may be accordingly written

$$\left[B_1\left(\frac{T_g-T_l}{H}\right)-\frac{1}{W_c}^n-Z\right]k=B_9\left(\frac{W_l}{\mu}\right)^{-m}$$
 (23)

· A logarithmic plot of the factor

$$\left[0.37 \left(\frac{T_{s}-T_{t}}{H}\right) - \frac{1}{W_{c}^{0.94}} - 0.12\right]k$$

against  $W_t/\mu$ , in which the absolute value of the slope of straight line through the data will be equal to the exposition, is shown in figure 28 for five different coolants. having the same slope are drawn through the data for coolant and the value of the exponent m is thus estable as 0.26. This value of the exponent m, which is estable from selected data, will be verified by the slope of the through all the data in the final correlation plot base equation (6).

Exponent s on Prandtl number Pr.—The value of exponent s on the Prandtl number Pr was determined data for a constant value of  $W_d/\mu$ , which permits the retion of equation (6) to

$$\left[B_1\left(\frac{T_{\kappa}-T_l}{H}\right)-\frac{1}{W_{\epsilon}^n}-Z\right]k=B_{10}(Pr)^{-s}$$

The slope of the line determined by a logarithmic pl  $\left[0.37 \left(\frac{T_s - T_l}{H}\right) - \frac{1}{W_e^{0.94}} - 0.12\right] k$  against Pr would the tablish the value of the exponent s.

Following a procedure similar to that used in the I temperature correlation, values of the factor

$$\left[0.37\left(\frac{T_g-T_i}{H}\right)-\frac{1}{W_0^{0.94}}-0.12\right]k$$

are obtained from figure 28 for each of the coolants at a constant value of  $W_t/\mu$  and are then cross-plotted against the Prandtl number Pr for the different coolants. Data so obtained from a cross plot of figure 28 at a constant value of  $W_t/\mu$  equal to 55,000 are shown in figure 29 and the slope of the resulting line establishes the value of the exponent s on the Prandtl number Pr as 0.38.

# FINAL CORRELATION

Final correlation with coolant-flow factor  $W_t/\mu$  as independent variable.—The final correlation based on equation (6), which is obtained by plotting the factor

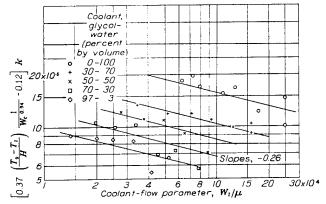


FIGURE 28.—Determination of exponent m on coolant-flow parameter  $W_I/\mu$  for coolant-heat-rejection correlation from variation of  $\left[0.37\left(\frac{T_{\ell}-T_I}{H}\right)-\frac{1}{W_{\mathfrak{g}}^{0.94}}-0.12\right]k$  with  $\frac{W_I}{\mu}$  for various coolants. Average coolant temperature, 245° F.

$$\left[0.37 \left(\frac{T_{\rm g} - T_{\rm f}}{H}\right) - \frac{1}{W_{\rm c}^{0.94}} - 0.12\right] k(Pr)^{0.38}$$

against  $W_t/\mu$  on logarithmic coordinates for all the dat presented in figure 30. Although the data points seconsiderably, the variation in terms of the coolant rejection is not excessive. Dashed lines representively variation in heat rejection of  $\pm 5$  percent are drawn or figure and practically all the data are seen to fall we these limits. The effect of boiling of the coolant on

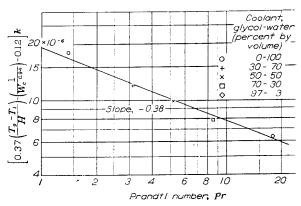


Figure 29.—Determination of exponents on Prandtl number Pr for coolant-heat rejection lation from variation of  $\left[0.37\left(\frac{T_s-T_l}{lI}\right)-\frac{1}{W_c^{0.94}}-0.12\right]k$  with Pr as obtain cross plot of figure 28 at value of  $\frac{W_l}{d}$  of 55,000. Average coolant temperature, 245°

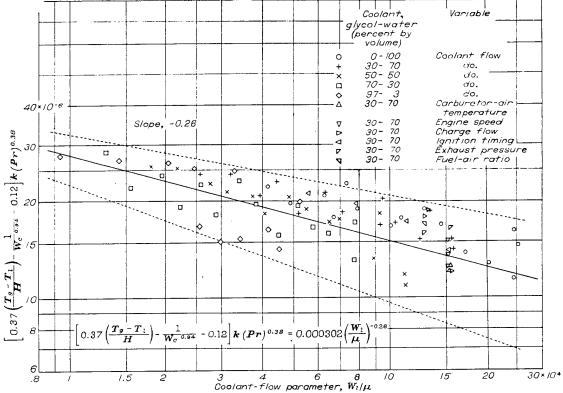


Figure 30.—Final correlation of coolant heat rejections based on equation (6).

correlation, which, as illustrated in figure 17, had a small influence on the cylinder-head-temperature correlation, could not be detected within the accuracy of the data. Because the region in which boiling of the coolant occurred was probably limited to a relatively small area of the cylinder head, it would be expected to have a much smaller effect on the total coolant heat rejections than on the cylinder-head temperatures.

The value of the exponent m, which is equal to the absolute value of the slope of the line through the data, is, as previously determined, equal to 0.26, and the value of the constant  $B_3$ , found by the substitution of the values of the coordinates of any point on the line into the correlation equation, is equal to 0.000302. The final equation is accordingly written

$$\left[0.37 {T_{\rm g}-T_{\rm l} \choose H} - {1 \over W_{\rm g}^{0.94}} - 0.12\right] k(Pr)^{0.38} = 0.000302 {V_{\rm g}^{0.94}}$$

and will apply over the range of engine operating con and coolant conditions listed in table I.

Final correlation with charge flow  $W_c$  as independent variable.—Correlation of the test data based on equation which is obtained by plotting the factor

$$0.37 \left(\frac{T_{\rm g}-T_{\rm t}}{H}\right) - \left(\frac{0.000302}{W_{\rm t}^{0.26}}\right) \left(\frac{\mu^{0.26}}{k(Pr)^{0.38}}\right) - 0.12$$

against  $W_c$  on logarithmic coordinates, is present figure 31. A straight line with a slope of -0.94, we equal to the value of the exponent on  $W_c$  previously mined, is drawn through the data. When equation

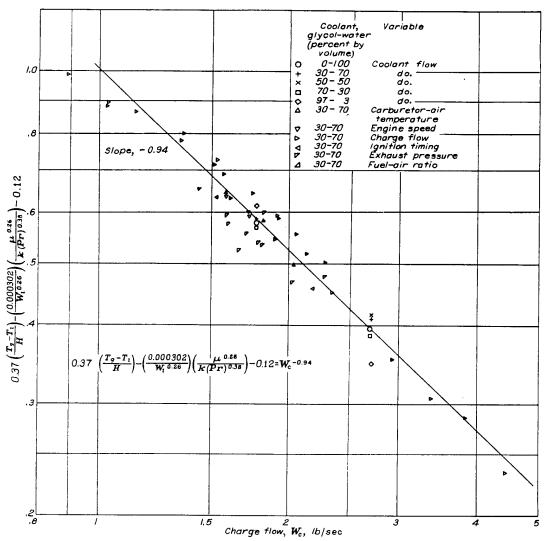


FIGURE 31.—Final correlation of coolant heat rejections based on equation (7).

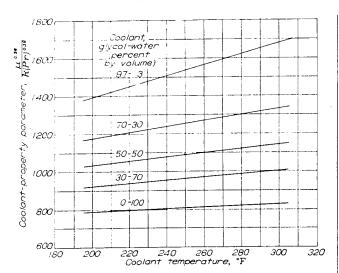


Figure 32.—Variation with coolant temperature of coolant-property parameter  $\mu^{0.28/\kappa}(P\tau)^{0.98}$  for coolant-heat-rejection correlation.

rearranged in accordance with this plot, the following form is obtained:

$$0.37 \left(\frac{T_s - T_t}{II}\right) - \left(\frac{0.000302}{W_t^{0.26}}\right) \left(\frac{\mu^{0.26}}{k(Pr)^{0.38}}\right) - 0.12 = W_c^{-0.94}$$
(26)

In order to facilitate the computation of coolant heat rejections by means of this equation, values of the coolant-property parameter  $\mu^{0.26}/k$  (Pr) <sup>0.38</sup> are presented in figure 32 for various ethylene glycol—water solutions over a range of coolant temperatures.

# USE OF CORRELATION EQUATIONS

In order to illustrate the use of the correlation equations, the following example is presented:

The maximum cylinder-head temperature between the exhaust valves and the coolant heat rejection are to be determined for the following conditions:

Engine charge flow (air plus fuel), lb/sec	3. 0
Engine speed, rpm.	3000
Fuel-air ratio	0.095
Carburetor-inlet air temperature, °F	
Exhaust pressure, in. Hg absolute	40. 0
Ignition timing (exhaust spark plugs), deg B.T.C	40
Accumulated engine running time, hrOv	er 100
Coolant flow, lb/sec	
Average coolant temperature, °F	
Coolant composition, ethylene glycol-water (percent by	
volume)	30 - 70

Determination of maximum cylinder-head temperature.— The average cylinder-head temperature is first evaluated from equation (18) and the maximum cylinder-head temperature then determined from figure 18. Although either equation (17) or (18) may be used to evaluate the average cylinder-head temperature, the grouping of the coolantproperty terms of equation (18) results in greater convenience of application.

The dry inlet-manifold temperature is computed from the carburetor-air temperature and the engine speed by means of equation (9) as follows:

$$T_m = T_c + 25.28 \left(\frac{N}{1000}\right)^2$$
$$= 60 + 25.28 \left(\frac{3000}{1000}\right)^2$$
$$= 288^{\circ} \text{ F}$$

In order to determine the effective cylinder-gas temperature  $T_s$ , a value of  $T_s$  is first determined from figure 6 as 1069° F for a fuel-air ratio of 0.095, an exhaust pressure of 40 inches of mercury absolute, standard ignition timing, and a dry inlet-manifold temperature of 80° F. The correction for an ignition timing of 40° B.T.C. is obtained from figure 7 as 24° F. For a dry inlet-manifold temperature of 288° F, the correction to  $T_s$  is determined from equation (11)

$$\Delta T_g = 0.25 (T_m - 80)$$
  
= 0.25 (288 - 80)  
= 52° F

The value of  $T_s$  is then determined by algebraically adding the corrections for ignition timing and manifold temperature to the value obtained from figure 6.

$$T_{\rm g} = 1069 + 24 + 52 = 1145^{\circ} \, \text{F}$$

Because the variation of Z with engine running time was shown to be constant at a value of 0.13 (fig. 11) for any engine running time over 100 hours, this constant value is used. The coolant-property parameter  $\left(\frac{\mu^{0.48}}{k(Pr)^{0.33}}\right)$  is determined from figure 16 for the specified coolant and coolant temperature as equal to 164.

Substitution of the values of the various parameters into equation (18)

$$\left( \frac{T_{\rm g} - T_{\rm h}}{T_{\rm h} - T_{\rm i}} \right) \left[ \left( \frac{0.00163}{W_{\rm i}^{0.48}} \right) \left( \frac{\mu^{0.48}}{k(Pr)^{0.33}} \right) + Z \right] = W_{\rm e}^{-0.60}$$
 (18)

gives the following:

$$\frac{\left(\frac{1145 - T_h}{T_h - 250}\right) \left[\left(\frac{0.00163}{30^{0.48}}\right) (164) + 0.13\right] = 3.0^{-0.60}}{2.839 T_h + T_h = 1145 + 2.839 \times 250}$$

$$T_h = 483^{\circ} \text{ F}$$

For this value of the average cylinder-head temperature, the maximum cylinder-head temperature between the exhaust valves is found to be 499° F (fig. 18).

Determination of coolant heat rejection. As for the cylinder-head temperature, the determination of the coolant heat rejection is most conveniently accomplished by using the equation in which the charge flow is the separated variable; equation (26) will therefore be used.

As previously calculated for the cylinder-head-temperature determination, the dry inlet-manifold temperature is 288° F. A value of  $T_s$  for a fuel-air ratio of 0.095, an exhaust pressure of 40 inches of mercury absolute, standard ignition timing, and a dry inlet-manifold temperature of 80° F is first determined from figure 23 as 716° F. The correction for an ignition timing of 40° B.T.C. is obtained from figure 24 as 14° F. For a dry inlet-manifold temperature of 288° F, the correction to  $T_s$  is determined from equation (20).

$$\Delta T_g = 0.30 \ (T_m - 80)$$
  
=  $0.30 \ (288 - 80)$   
=  $62.4 \ F$ 

The value of  $T_s$  is then determined by algebraically adding the corrections for ignition timing and manifold temperature to the value obtained from figure 23.

$$T_{\rm g} = 716 + 14 + 62.4 = 792.4^{\circ} \text{ F}$$

The coolant-property parameter  $\left(\frac{\mu^{0.26}}{k(Pr)^{0.38}}\right)$  is determined from figure 32 for the specified coolant and coolant temperature and is equal to 964.

Substitution of the values of the various parameters into equation (26)

$$0.37 \left(\frac{T_{\rm e} - T_{\rm f}}{H}\right) - \left(\frac{0.000302}{W_{\rm f}^{0.26}}\right) \left(\frac{\mu^{0.26}}{k(P_{\rm f})^{0.38}}\right) - 0.12 = W_{\rm e}^{-0.94} \quad (26)$$

gives the following:

$$0.37 \left(\frac{792.4 - 250}{H}\right) - \left(\frac{0.000302}{30^{0.26}}\right) (964) - 0.12 = 3.0^{-0.94}$$

$$\frac{200.7}{H} - 0.120 - 0.12 = 0.3561$$

$$H = 336.6 \text{ Btu per second}$$

# SUMMARY OF RESULTS

An analysis of the data obtained from multicylinder, liquid-cooled engines of 1710-cubic-inch displacement, which included power outputs from 275 to 1860 brake horsepower,

coolant flows from 50 to 320 gallons per minute, and coo composed of ethylene glycol—water mixtures varyin composition from 100 percent water to 97 percent ethy glycol and 3 percent water gave the following results:

- 1. The NACA correlation method, which is based of theory of heat transfer by normal forced convection, vided satisfactory correlation of both the cylindertemperature between the exhaust valves and the coheat rejection with the primary engine and coolant varifor a wide range of engine and coolant conditions.
- 2. The correlation method as applied herein perm the prediction of the cylinder-head temperature bet the exhaust valves within approximately  $\pm 12^{\circ}$  F at the coolant heat rejection with an accuracy of  $\pm 5$  per
- 3. Boiling of the coolant, which was encountered at se engine powers under certain conditions of coolant coolant temperature, and coolant composition, was f to have an effect on the cylinder-head temperatures, the ranges of variables covered in this investigation, ever, this boiling of the coolant did not seriously affect cylinder-head-temperature correlation and had no detection the heat-rejection correlation.

FLIGHT PROPULSION RESEARCH LABORATORY, NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS. CLEVELAND, OHIO, August 31, 1948.

# REFERENCES

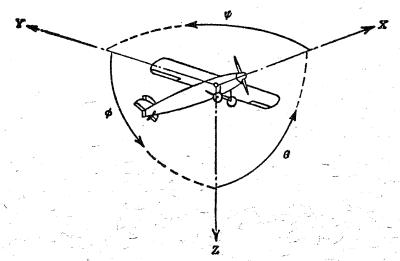
- Pinkel, Benjamin, Manganiello, Eugene J., and Bernardo, Ev Cylinder-Temperature Correlation of a Single-Cylinder L. Cooled Engine. NACA Rep. 853, 1946.
- Pinkel, Benjamin: Heat-Transfer Processes in Air-Cooled F. Cylinders. NACA Rep. 612, 1938.
- Povolny, John H., and Chelko, Louis J.: Cylinder-Head Tentures and Coolant Heat Rejections of a Multicylinder, L. Cooled Engine of 1710-Cubic-Inch Displacement. NAC. 1606, 1948.
- Cragoe, C. S.: Properties of Ethylene Glycol and Its Aq Solutions. Cooperative Fuel Res. Comm., CRC, July 19
- Manganiello, Eugene J., Valerino, Michael F., and Bell, E. Br High-Altitude Flight Cooling Investigation of a Radial Cooled Engine. NACA Rep. 873, 1947.
- Valerino Michael F., Kaufman, Samuel J., and Hughes, Richar Effect of Exhaust Pressure on the Cooling Characteristics Air-Cooled Engine. NACA TN 1221, 1947.
- Pinkel, Benjamin, and Ellerbrock, Herman H., Jr.: Correlation Cooling Data from an Air-Cooled Cylinder and Several Poylinder Engines. NACA Rep. 683, 1940.

TABLE I-SUMMARY OF ENGINE AND COOLANT CONDITIONS

Engine running time	(hr)		35, 9- 45, 8	153. 8-156. 5	2- 51.2 K-17-9		:	c	4.1	Á	2-33-2	1- 35		2.38.5	112,2,116,1	7. 5-151.4 7. 5-151.4	142, 2, 145, 8	5.5-185.5	1, 6-142, 2	138, 5-140, 6	5.8-148.5		÷	88,71 98,8	20.00	03.5-110.2			
	Heat rejec- tion	1			x 45		-			. K	8 E	× ×			-	± ±	. I+:	≊.	. <del></del>	ž ×	=======================================		×	35 : 	5i 3				
Correlation used in	Tem- H pera- re ture ti		<u>:</u> , :		× ×	:							.i	. ×	× :		×	 .e		×	× .			~			-		
Engine		. U2				:		 : U £				<u>-</u>	 :=			 	۵.	 = =	: :=				_				<u> </u>		:
Engine cookant- outlet Pressure	(lb'sq in. gage)	- 78 F	3 12	 R R	:£ :£	150	52	 	1 12	32	25	25	21:	35.	5	7.5	35	 	32	35	g S	28.0	35	12 1	2 5	12	35	- 35 - 35 - 35 - 36 - 36	2
Average coolant temper-	(4.E)	545	25.5	245	245	245	245	355	157	245	25.5	245	27.5	245	245	252	245	245	245	245	245	12	215	200	1 5	212	149-283	150-312	245
Coolant flow (gal/	i i	1255	250	255	5 5	988	28	270	9	008	9.8	300	3.2	8	<u> </u>	95	25	35	18	230	967.536	(A)	S(F3(8)	3.5	6.6	90.00	96	9.5	114-255
Coolant, ethylene glycol—water (percent by volume)	- Water	ļ	2,7,1											_				_		_			_						
	Glycol	!	88																					_				3 8 1 8	
Dry inlet- manifold temper-	##. (*F)	. E	256	194-316	168-345	208-34	(223-278	151-341	185-37	151-27	118-27	244-40	3 E	25.	= : = :	181	128	25	នា	22.5	7	1 24	25. E. #3.	254,31	32.	254, 31	_	75.5	<u> </u>
Ex- haust pres- sure (in.	age est inter			29-30			_		00 00	~			23-30	_	08-6K	٠.		10-52			_	~~		3 2			39-30		9. 2.
Ignition timing eg B.T.C.)	is is		888															_	_		_						_		
Ignition timing (deg B.T.	Ex- haust	# #		ਝ <b>ਲ</b> 	# # #	ž.	7.7	ੜ <i>ਹ</i>	· #	# 5	3.5	# i	7 F	**	75-51 	: # :	33	# #	#	# 2	\$ Z	· **	#	7	: #	8	# i	5 F	 
Carbure- tor-air temper-	(4.E)	85	88,	î8	ŹŹ	26.5	37-12	-29-170	-43-146	<u>=</u> ;	212	Ξ	2 2	52	<del>2</del> 2	2.53	<del></del> ;	\$ £	-2	61	2 15	2	Z	23	2 2	Ø	2:	2 75	55
Fuel-air ratio		0.050			5.5	•	980	0.050 1.050 1.050	.095	•	. 66 . 66 . 66		661-780					65.0											. 064-, 096
Charge flow (lb/sec)		1, 03-2, 47	1.35-2.76	1.70-2.18	2 1 2 1 3 3 1 3 1 3 1	1,35-1,60	7 T	25.5	5.3	21.	1, 75	1.58	3.2	유 하		1.43	3 5		2.38	<b>3</b> 3	1.00 1.05	2.21	1.7.2.2	1. 1. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2.	1.35	1, 78, 2, 71	2:	2, 39, 2, 94, 3, 40	25-15 E
Manifold pressure (in. Hg	absolute	24-48 21-54	: Zi	2 Q :	∓ ≅	<b>R</b> S	34-36	8 A	98-49	# F	77	31-40	3 <del>3</del> <del>3</del>	77-68 20-71	R \$	ş	9 9 8 8	9-98	40-47	30-41	? <del>1</del>	#	25 E	\$ 5 \$ 5	3	36, 50	æ:	42, 51, 60	24-53
Engine speed (rom)		2600 2600	98	2300-3200	1400-1500 1400-1500	2200-3200	0000 0000	9,50	3000	926-326	2000-3200	2000-3200	9,09	3000		(K)(S)	955	25.0	3000	3000	200	Sego	2600, 3000	SYNK) SORRI	2500, 3000	2600, 3000	998	3000	2000-3000
Engine power (bhp)		400-1150	553-1260	0±6-1998	130 630	029-019	#62-9X2	555-590	258-846	642-825	662-868	546-742	731-837	260 960	929-22	540-612	252-355 257 355	071-1750	944-994	592-682	1000	TOKK	2.5	3.5	280, 1180	180, 1160	587	1000, 1250, 1450	310-1200
Variable or type of test		Charge flow					Manifold temperature					Berry of a mostle	r derail tatto		Ignition timing	Exhaust pressure:					Coolant flow						Coolant temperature	Engine calibration	•

Engine equipped with aftercooler, AN-E-2 ethylene glycol.

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Positive directions of axes and angles (forces and moments) are shown by arrows

- Axis			Mome	nt abou	ıt axis	Angle		Velocities				
 Designation	Sym- bol	Force (parallel to axis) symbol	Designation	Sym- bol	Positive direction	Designa- tion	Sym- bol	Linear (compo- nent along axis)	Angular			
Longitudinal Lateral Normal	$X \\ Y \\ Z$	X Y Z	Rolling Pitching Yawing	L M N	$\begin{array}{c} Y \longrightarrow Z \\ Z \longrightarrow X \\ X \longrightarrow Y \end{array}$	Roll Pitch Yaw	φ θ ψ	u v w	P • Q			

Absolute coefficients of moment 
$$C_1 = \frac{L}{qbS}$$
  $C_m = \frac{M}{qcS}$   $C_n = \frac{N}{qbS}$  (rolling) (pitching) (yawing)

Angle of set of control surface (relative to neutral position), &. (Indicate surface by proper subscript.)

# 4. PROPELLER SYMBOLS

D	Diameter	P	Power, absolute coefficient $C_P = \frac{P}{an^3D^5}$
p	Geometric pitch	• •.	<i>p</i>
p/D	Pitch ratio	$\boldsymbol{C}$	Speed-power coefficient = $\sqrt[5]{\frac{\rho V^5}{Pn^2}}$
V'	Inflow velocity	<b>0.</b>	VPn2
$V_{\bullet}$	Slipstream velocity	η	Efficiency
T	Thrust, absolute coefficient $C_T = \frac{T}{\rho n^2 D^4}$	n	Revolutions per second, rps
		Φ	Effective helix angle = $\tan^{-1} \left( \frac{V}{2\pi rn} \right)$
Q	Torque, absolute coefficient $C_Q = \frac{Q}{\rho n^2 D^5}$		

# 5. NUMERICAL RELATIONS

1  hp = 76.04  kg-m/s = 550  ft-lb/sec	1 lb=0.4536 kg
1 metric horsepower=0.9863 hp	1 kg=2.2046 lb
1 mph=0.4470 mps	 1  mi = 1,609.35  m = 5,280  ft
1  mps = 2.2369  mph	 1  m = 3.2808  ft